

# Ablation and Heating During Atmospheric Entry and Its Effect on Airburst Risk

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# Heating and Ablation in Threat Assessment

*Asteroid Entry Equation of Motion* \*

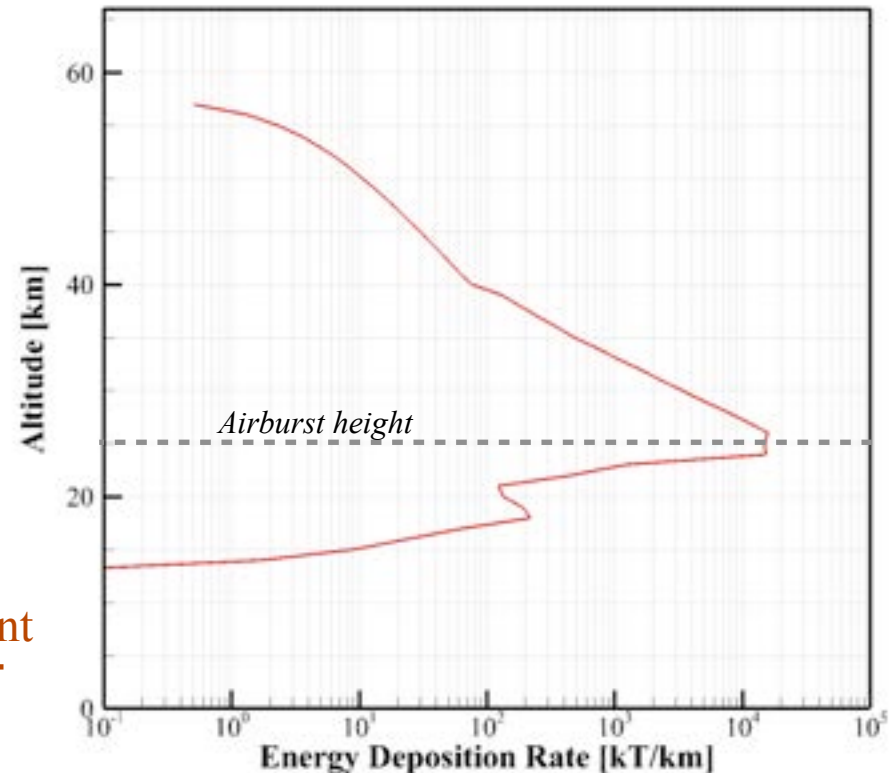
$$dv = \left( -\frac{1}{2} C_d \rho_A v^2 A / m + g \sin \theta \right) dt$$

$$d\theta = \left( \frac{v}{R_E + h} + \frac{g}{v} \right) \cos \theta dt$$

$$dm = -\frac{1}{2} \rho_A v^3 A \sigma_{ab} dt$$



$$\sigma_{ab} = \frac{C_H}{Q^*} \quad \begin{array}{l} \text{Heat Transfer Coefficient} \\ \hline \text{Heat of Ablation} \end{array}$$



# Heating and Ablation in Threat Assessment

*Asteroid Entry Equation of Motion* \*

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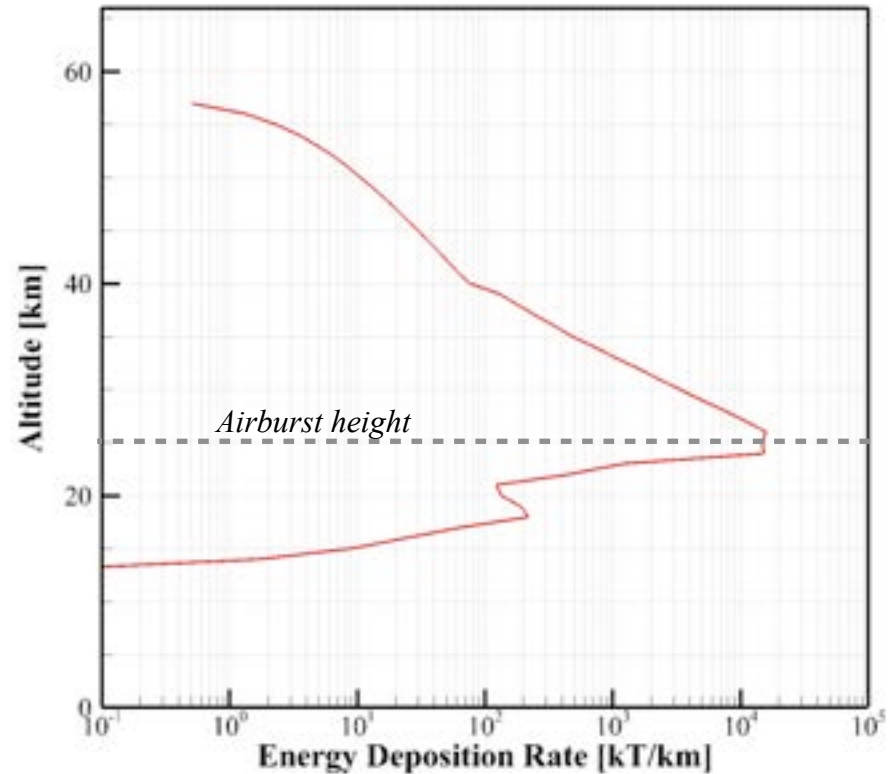
$$dm = -\frac{1}{2} \rho_A v^3 A \sigma_{ab} dt$$

$$\sigma_{ab} = \frac{C_H}{Q^*}$$

$$= 0.1$$

$$= 8 \text{ MJ/kg}$$

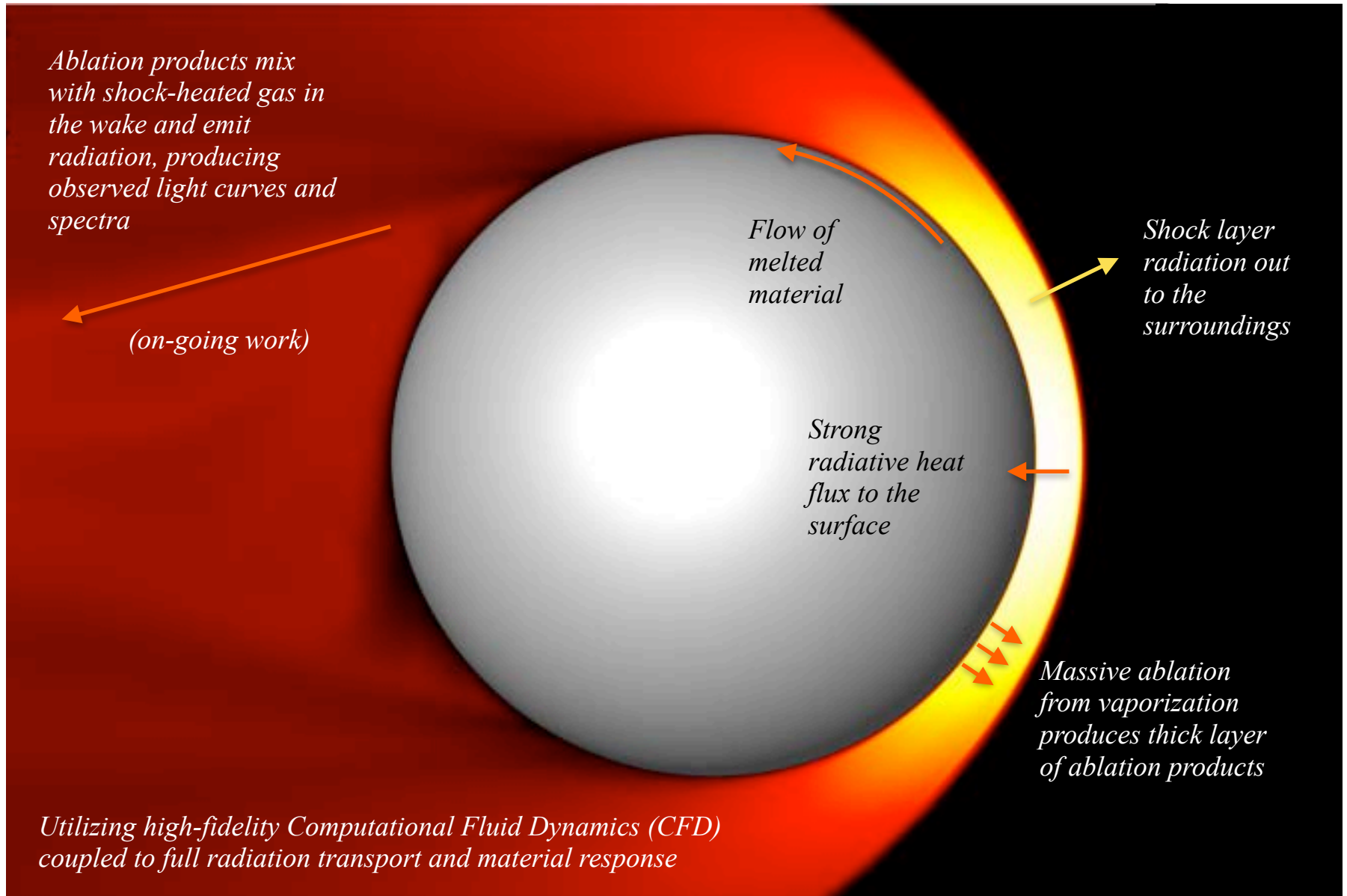
*Nominal Values*

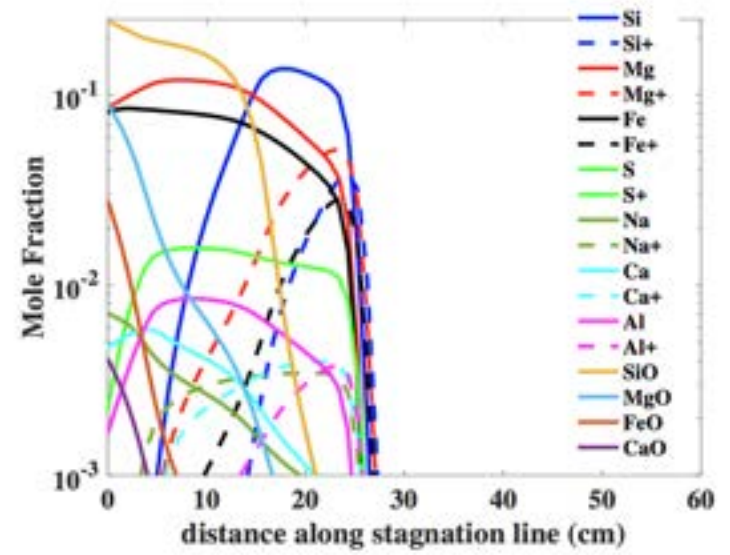
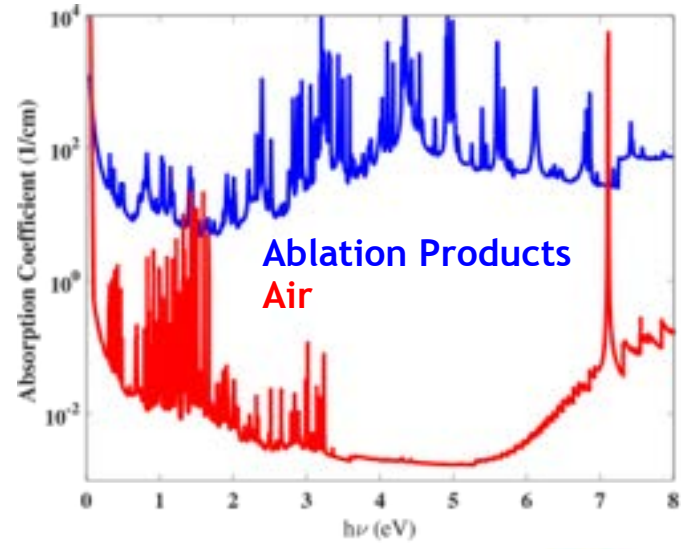
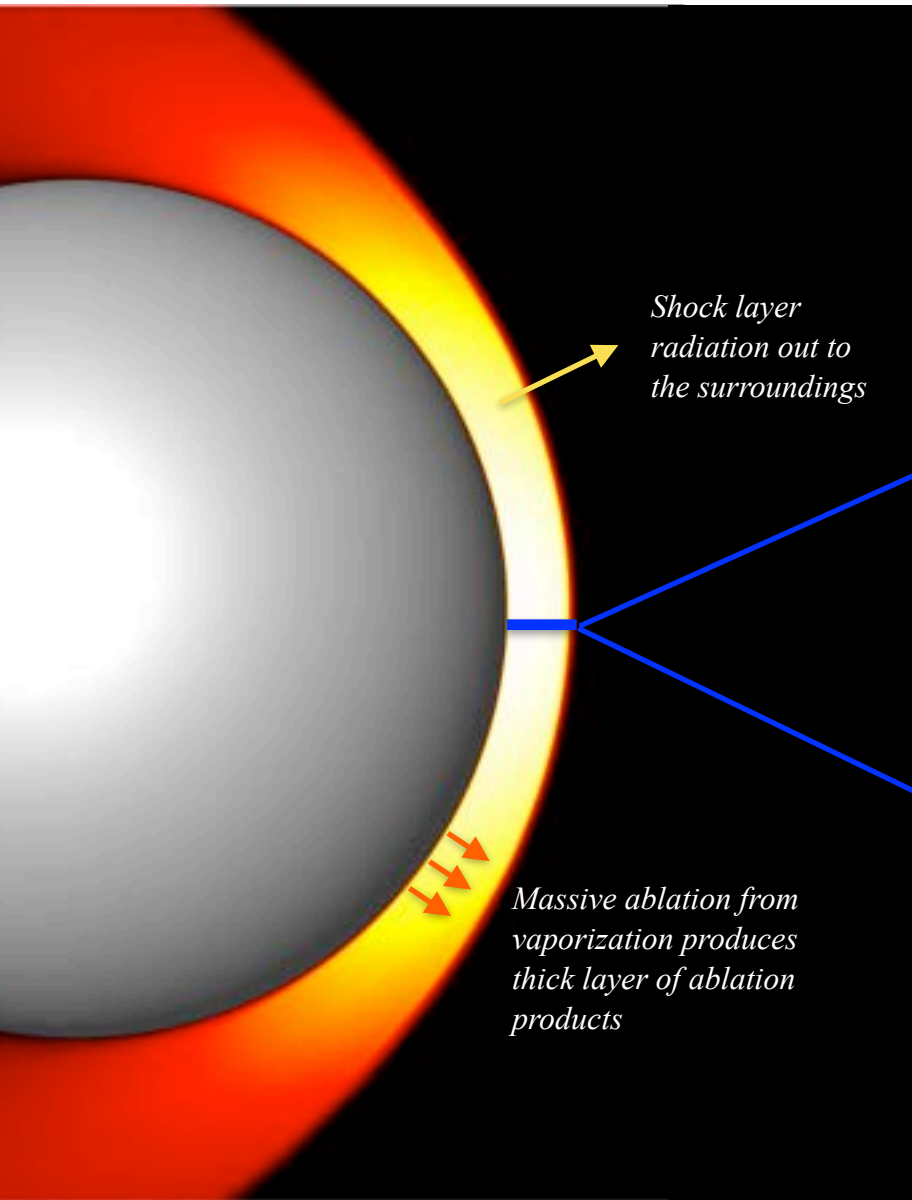


*NASA Asteroid Threat Assessment Project working to improve models for these phenomena*

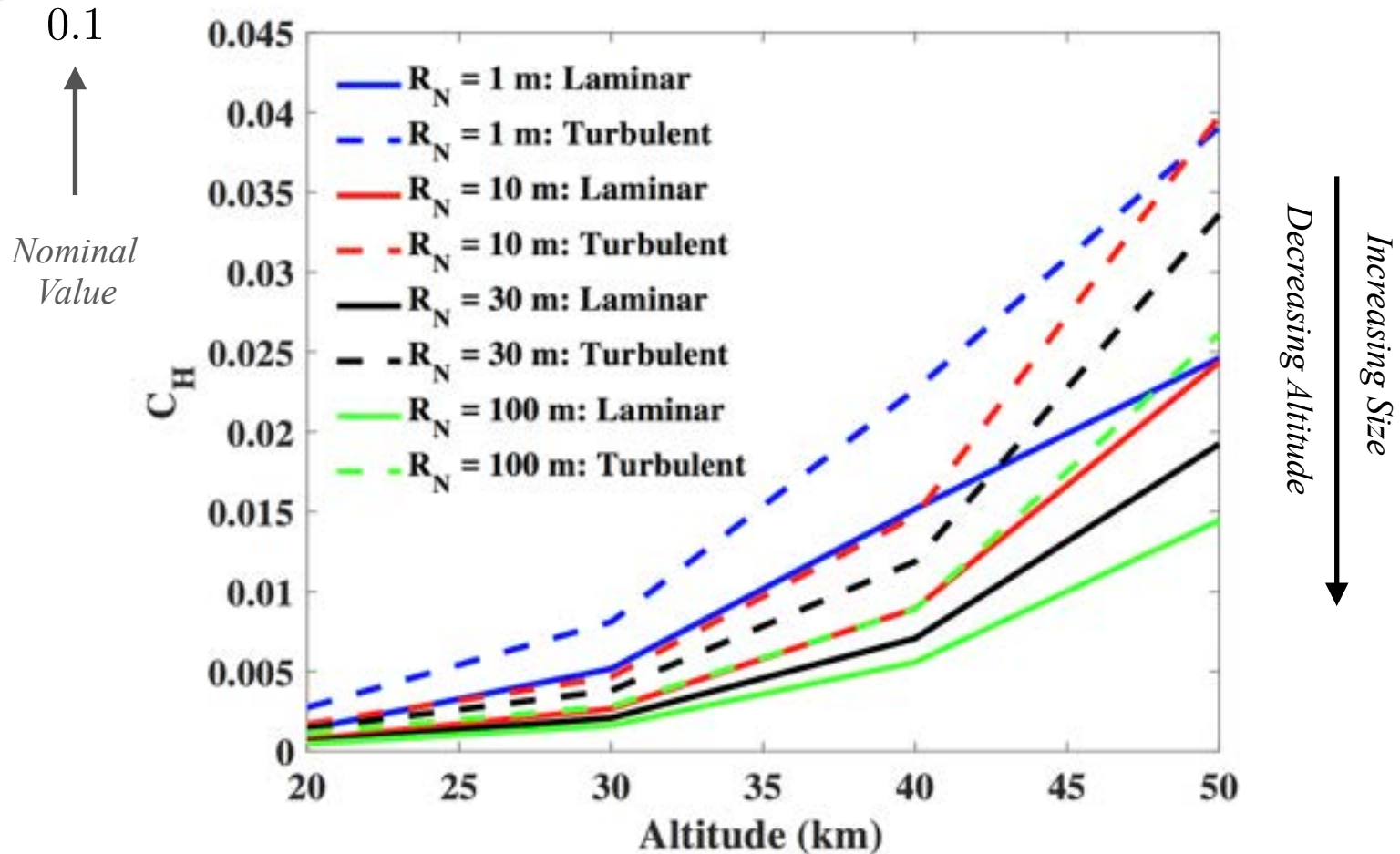
\* Wheeler et al., 2017

# Asteroid Entry Environment





\* Johnston and Stern., 2017

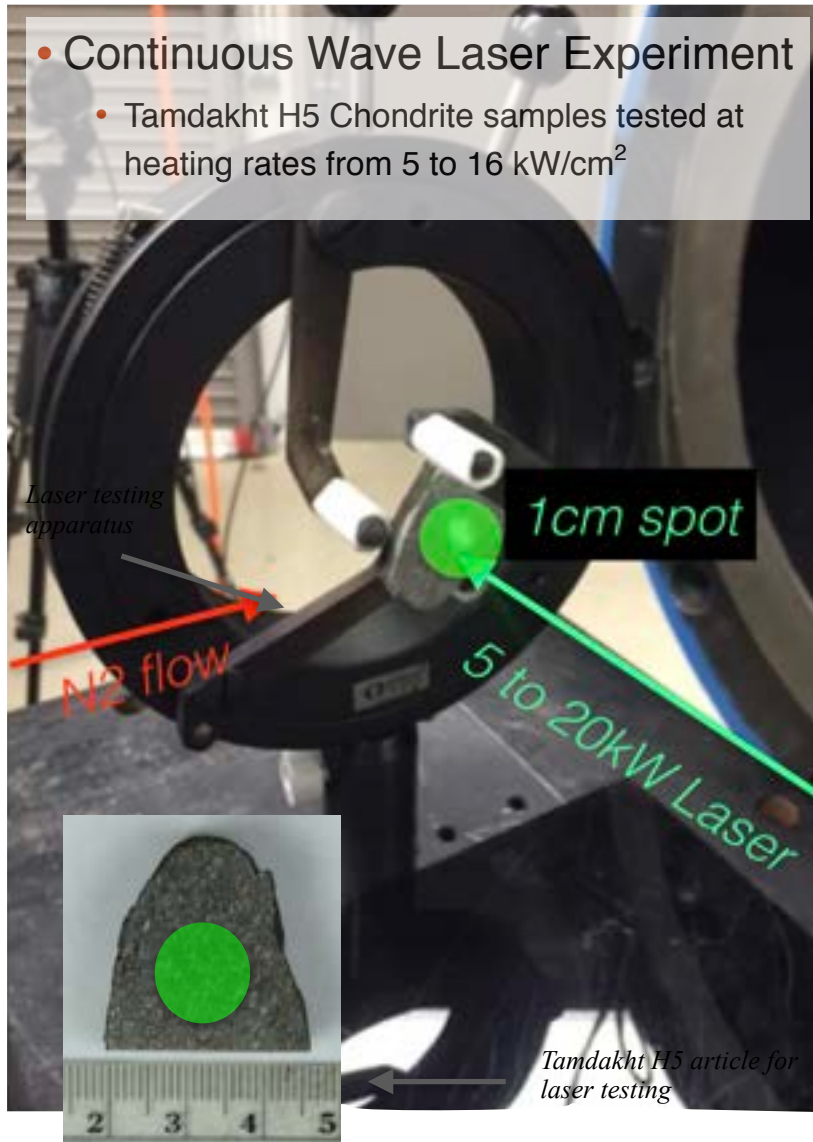


- Fully coupled radiation and ablation results reduces the heat transfer coefficient by nearly **two** orders of magnitude in some cases
- C. Johnston, E. C. Stern, L. F. Wheeler, "Radiative Heating of Large Meteoroids During Atmospheric Entry," *Icarus*, Vol. 31, p. 25-44, doi:10.1016/j.icarus.2018.02.026

# Meteoroid Ablation Experiments

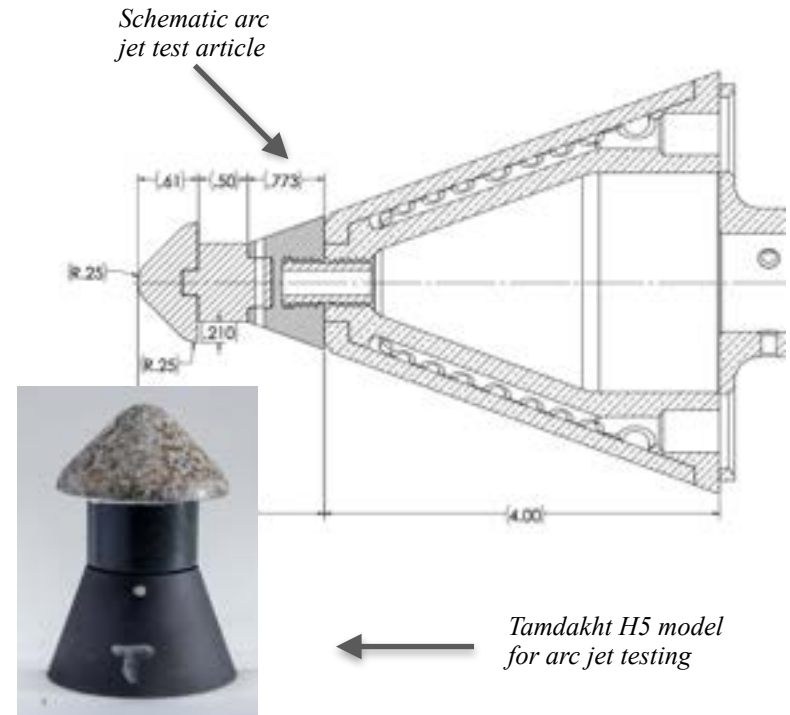
- Continuous Wave Laser Experiment

- Tamdakht H5 Chondrite samples tested at heating rates from 5 to 16 kW/cm<sup>2</sup>

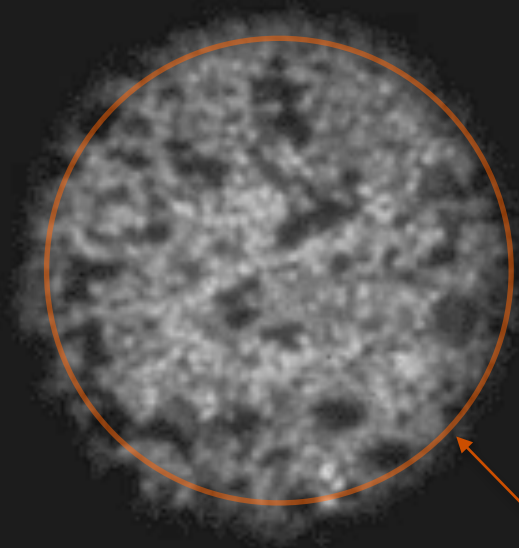


- Arc Jet Experiment

- Heating rates (~4 kW/cm<sup>2</sup>) produced in the experiment comparable to 30m asteroid at 20 km/s at 65km altitude
- Machined sphere-cone model allows for high-fidelity simulation of the test environment and material response



*Tamdakht H5 Chondrite*



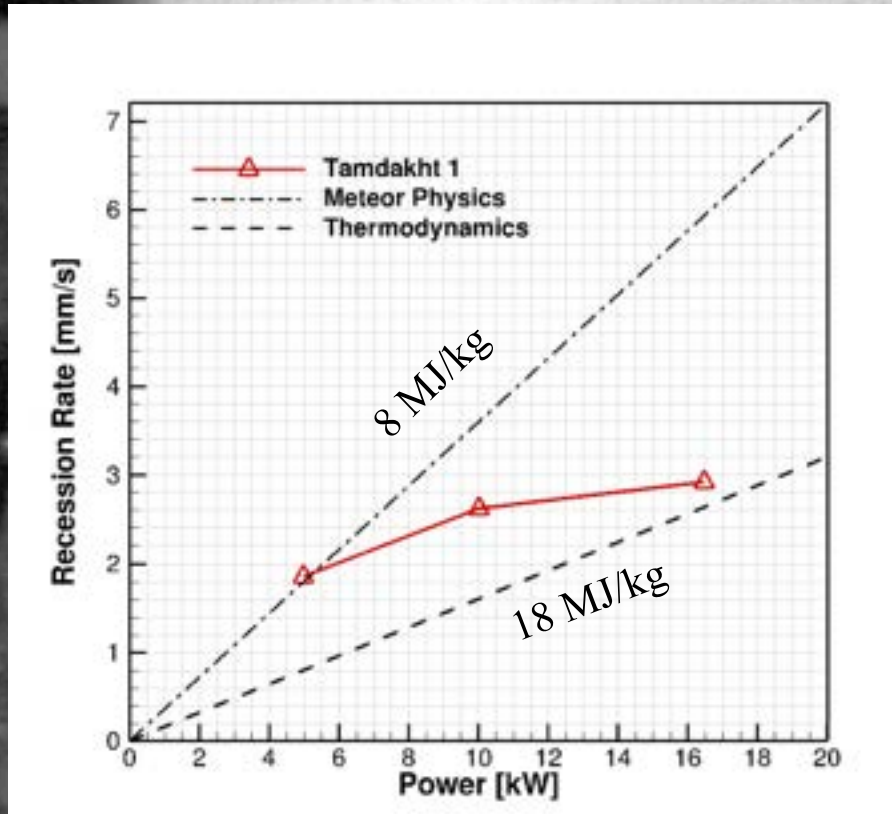
*5 kW/cm<sup>2</sup> Laser Spot*



*High-speed video showing boiling meteorite surface*



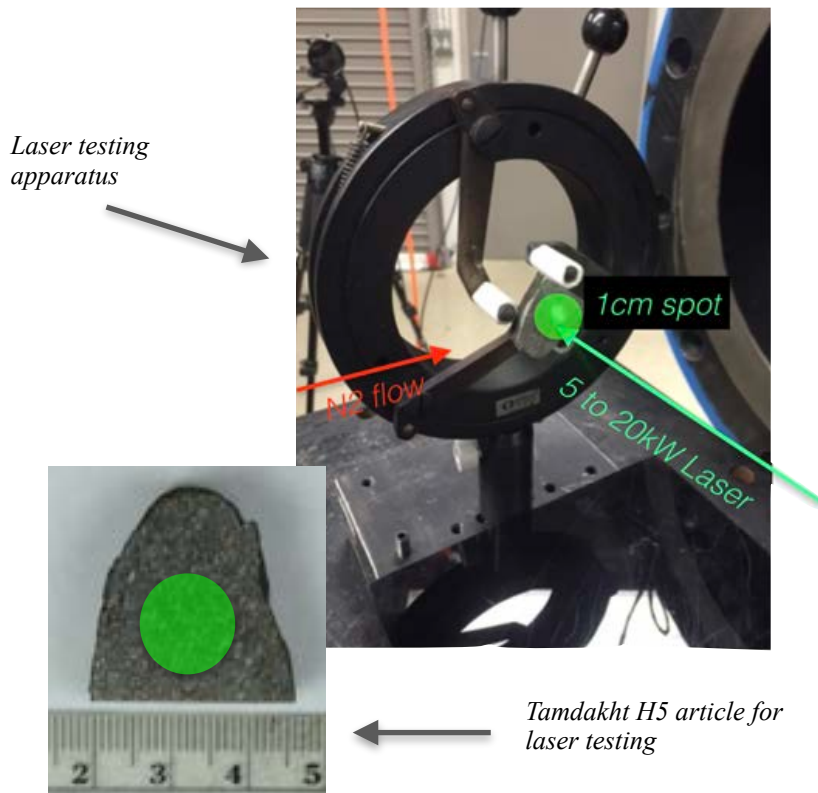
# Laser Experiment Findings



- At low heat flux, effective heat of ablation value close to canonical value of 8 MJ./kg
- Reduction in ablative efficiency at high heat fluxes attributed primarily to radiation blockage from ablation products

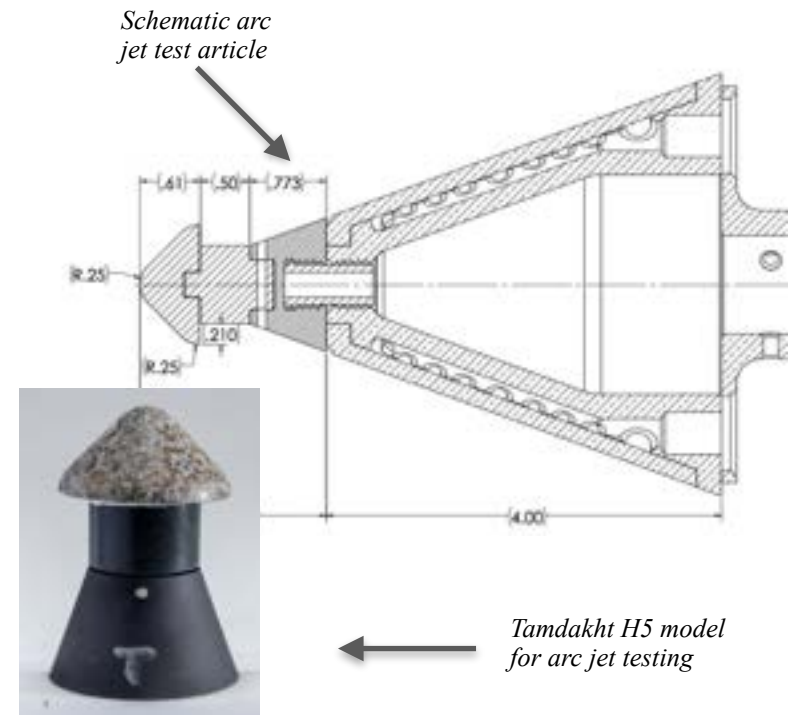
## • Continuous Wave Laser Experiment

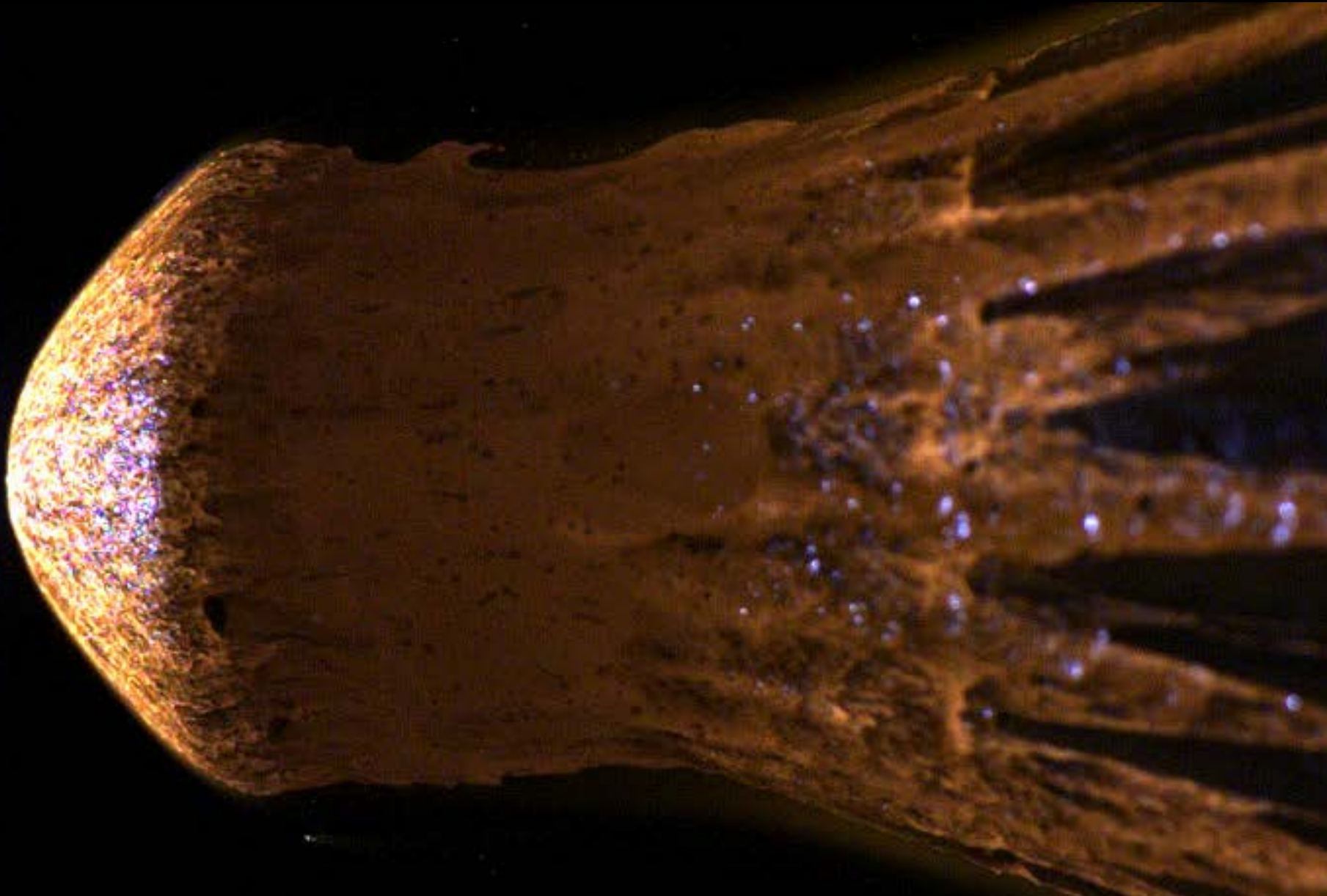
- Source of heating is radiation, which is the dominant source of heating for large meteoroids
- Tamdakht H5 Chondrite samples tested at heating rates from 5 to 16 kW/cm<sup>2</sup>



## • Arc Jet Experiment

- Heating rates (~4 kW/cm<sup>2</sup>) produced in the experiment comparable to 30m asteroid at 20 km/s at 65km altitude
- Machined sphere-cone model allows for high-fidelity simulation of the test environment and material response

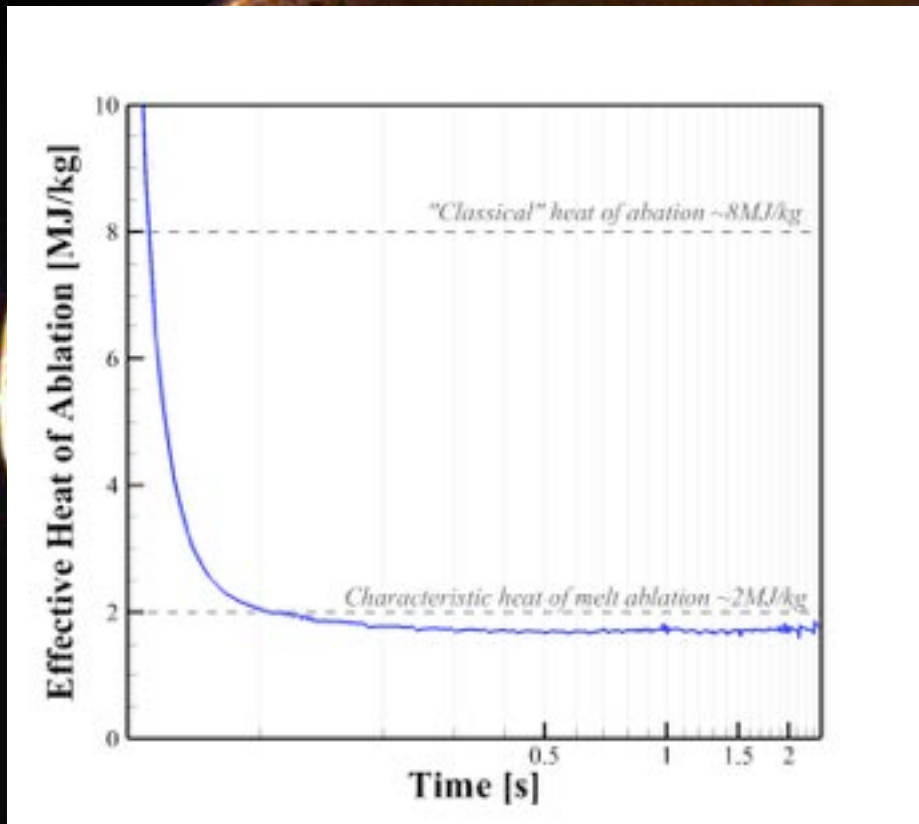




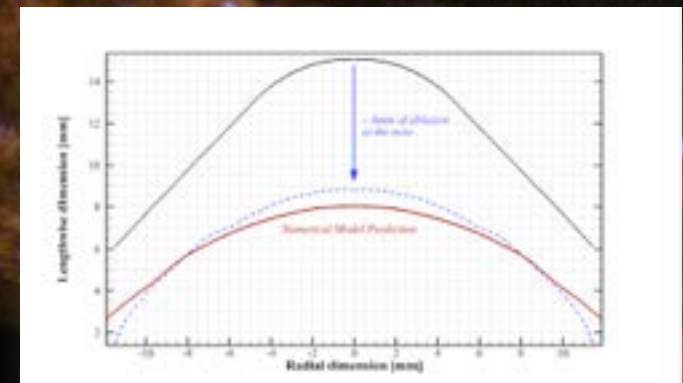
*High-speed video from arc jet experiment showing widespread melt flow*

# Arc Jet Experiment

## Findings



- Effective heat of ablation ( $Q^*$ ) from the experiment  $\sim 2$  MJ/kg
- Heat is well below the canonical value of 8 MJ/kg for chondrite vaporization
  - Indicates we are in a *melt* dominated regime



P. Agrawal, E. C. Stern, J. O. Arnold, Y-K. Chen, P. Jenniskens, "Arcjet Ablation of Stony and Iron Meteorites," AIAA Paper, Atlanta, Georgia, June 2018

# Effect of Ablation Parameter on Energy Deposition

Nominal Value

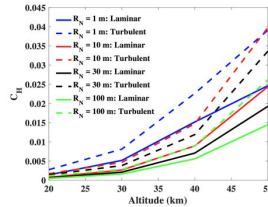
Range based on preceding analysis

$C_H$

0.1

0.001

0.04

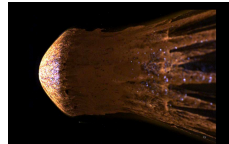


$Q^*$

8.0

1.8

8.0

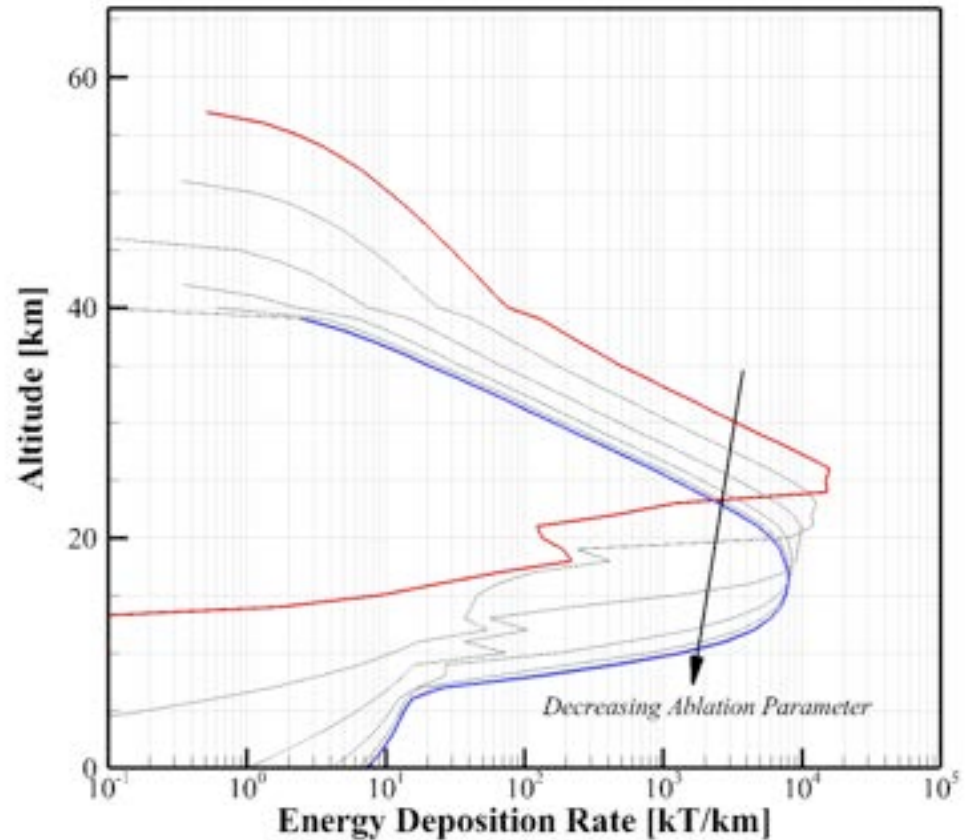


$$\sigma_{ab} = \frac{C_H}{Q^*}$$

$1.25 \times 10^{-10}$

$2.20 \times 10^{-8}$

100m diameter, 20km/s velocity, 83Mt

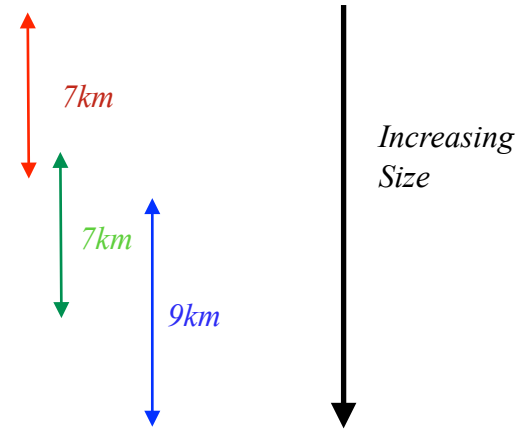
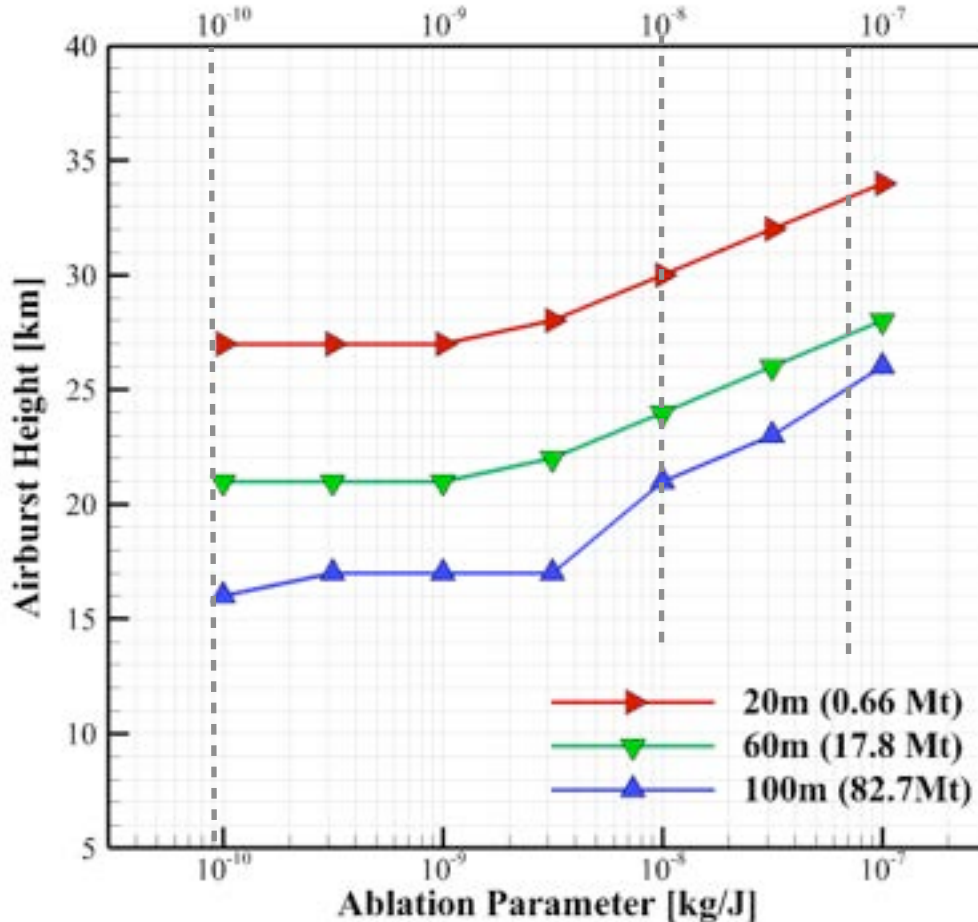


# Effect of Ablation Parameter on Energy Deposition

*Strongly coupled  
and vaporization  
dominated*

*Nominal*

*Uncoupled  
and Melt  
Dominated*



For 100m impactor, 9km burst height difference corresponds to 25km increase in 4psi blast footprint radius (using Glasstone and Dolan)

# Conclusions

- Coupled Fluid Dynamics-Ablation-Radiation calculations show significant reduction in heating over canonical value, particularly at larger sizes relevant to planetary defense
- Ground test experiments yielding insight into ablation phenomena, and being used to develop and validate numerical models
- Bias in ablation parameter toward the low-end results in lower altitude airburst, and therefore larger ground damage footprints

# Acknowledgments

- Work was performed under the Asteroid Threat Assessment Project, administered by the NASA Planetary Defense Coordination Officer, Lindley Johnson
- The NASA Interaction Heating Facility (IHF) Team is gratefully acknowledged for supporting the arc jet test
- The Air Force Research Laboratory Laser Hardened Material Evaluation Laboratory is gratefully acknowledged for supporting the laser testing
- Thanks to Greg Gonzalez and Val Kasvin for machining the models for the experiments