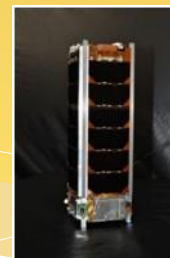
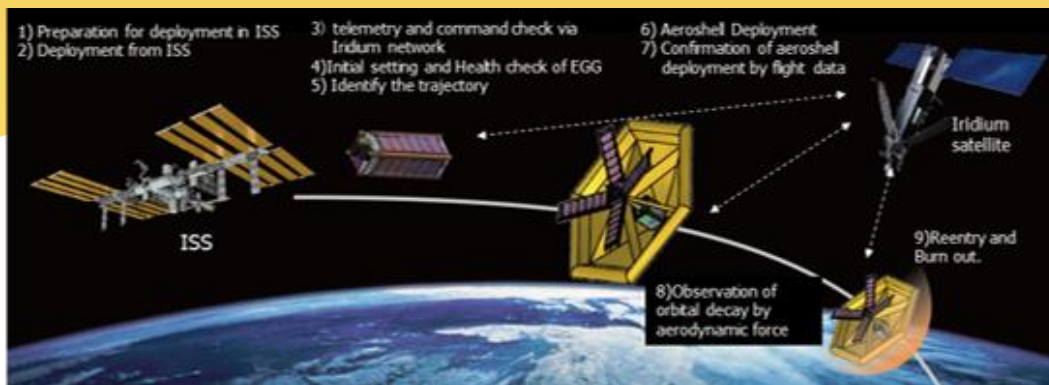
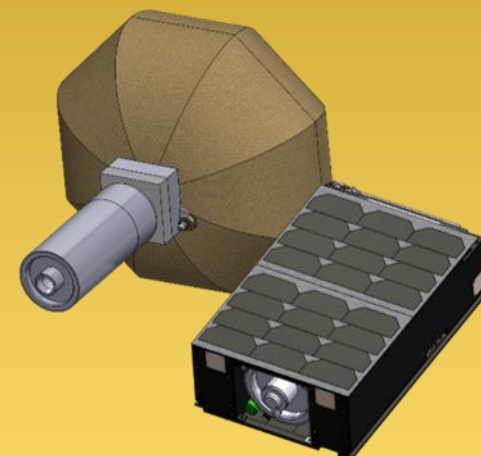
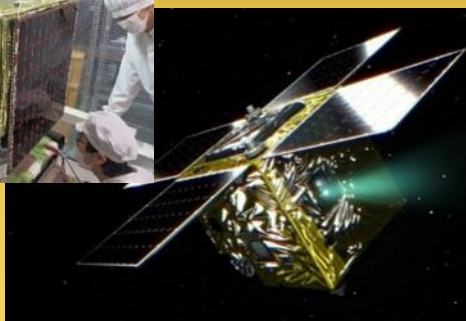
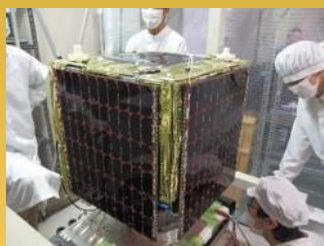


Short Course : Small Satellite

# SmallSat Efforts in JAPAN for future planetary exploration.

Kazuhiko Yamada (JAXA/ISAS)



# Self introduction

Name: Kazuhiko Yamada

Affiliation : JAXA/ISAS

My major : Atmospheric entry technology

“especially”,

**deployable membrane aeroshell  
for atmospheric-entry capsule.**

I have led Japanese activity related to the research and development related to deployable aeroshell since I was a student in 2000.

2004 : 1<sup>st</sup> balloon drop test

2009 : 2<sup>nd</sup> balloon drop test

2012 : Suborbital reentry demonstration using sounding rocket

**2017 : Nanosatellite EGG with deployable aeroshell**



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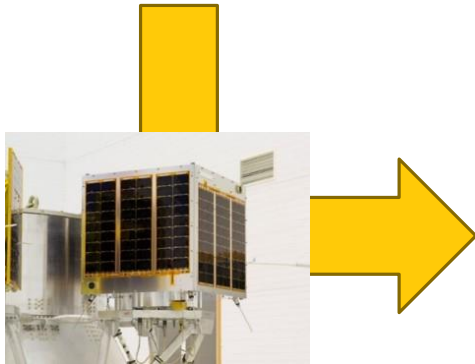
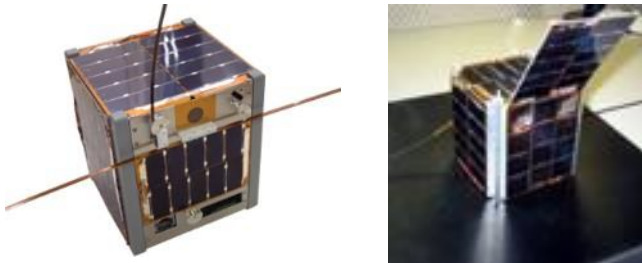
- Background
  - Potential of Small satellite.
- PROCYON's challenge to planetary exploration
- Next challenge using SLS opportunity
  - EQUULEUS
  - OMOTENASHI
- Innovation of Atmospheric-entry technology
  - Deployable aeroshell for small sat
  - EGG → BEAK → SPUR
- Future vision and Summery
  - New exploration world with small-sat tech.

# Start of Small-satellite Era

**1999 : Cube-sat spec. development**

**2003 : First launches of 1U-Cubesat s**

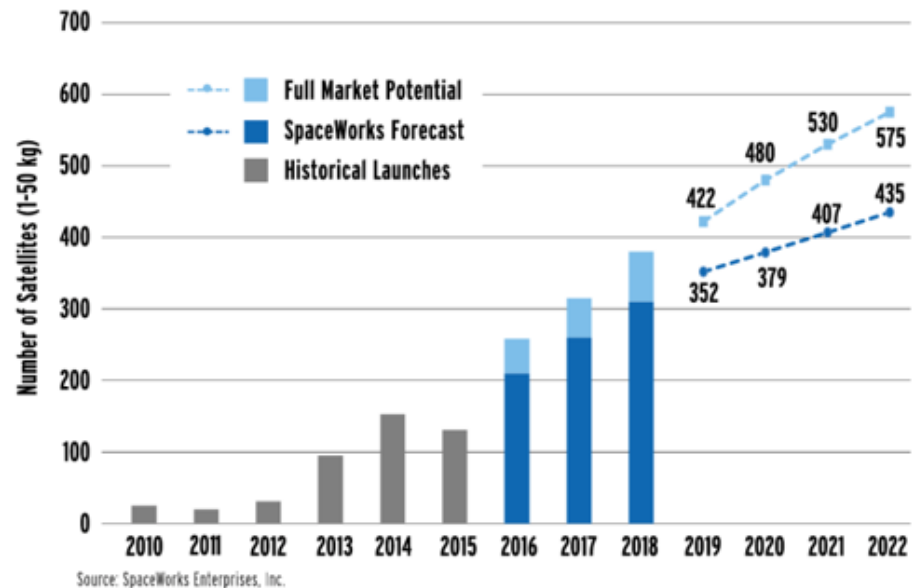
(From SpaceNews Magazine)



**HODOYOSHI series (2014-)**

## Nano/microsatellite launch history and forecast

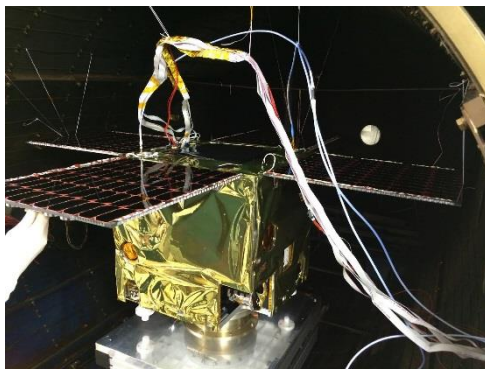
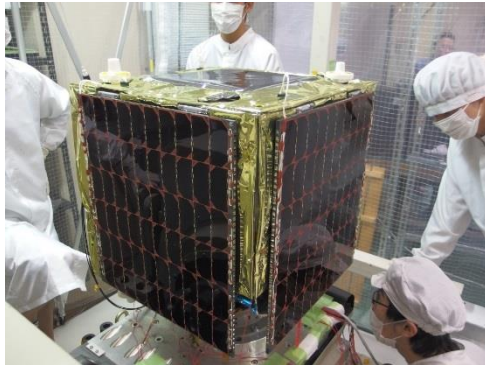
Projections based on announced and future plans of developers and programs indicate as many as 3,000 nano/microsatellites will require a launch from 2016 through 2022.



**Many small-satellites have been launched and operated in LEO.  
It is time to challenge to next field, that is, “planetary exploration”**

# PROCYON's challenge

**PROCYON : The first interplanetary Full-scale Micro-spacecraft**  
**PROCYON was developed by The University of Tokyo with collaboration team of many universities and JAXA/ISAS.**



## PROCYON spec.

Size:	0.55m x 0.55m x 0.67m + 4 SAPs
Weight:	<70kg (wet), payload(camera) 10kg.
Power	>240W SAP and Li-ion BAT 5.3Ahr
AOCS	4RW and 3-axis FOG, STT, NSAS <0.002deg/s, <0.01deg
Propulsion	Cold gas RCS (22mN, 24s ISP) and Ion propulsion(0.3mN, 1000s ISP) 2.5kg Xenon installed
Communication	X-band, DDOR (HGAx1, MGAx1, uplink LGAx2, downlink LGA x2), >15W

**PROCYON was developed in only 14months utilizing the heritage of Cube-sat and HODOYOSHI-series and launched in Dec. 2014.**

# PROCYON's achievement

**Demonstration of deep space bus system**  
including deep space communication and trajectory  
guidance/navigation/control → **success!**

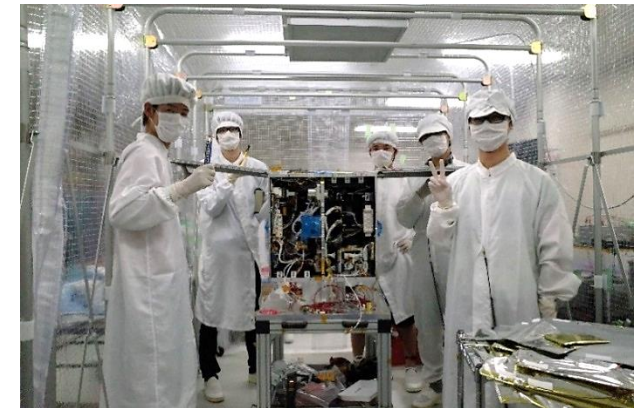
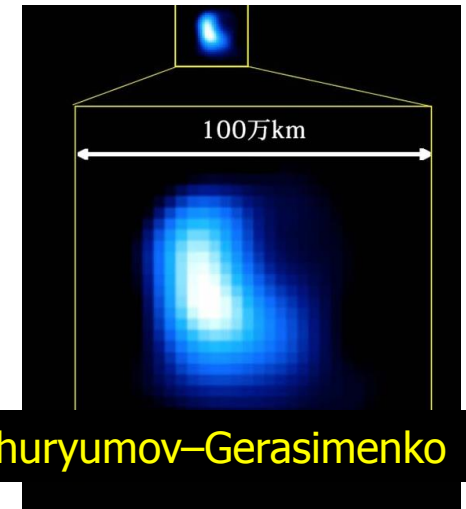
## Scientific mission

(geocorona observation and observation of hydrogen  
emission around 67P/C-G) → **success!**

All the mission were successful except for:

- **long-time deep space maneuver** by the ion thruster
- *actual* **asteroid flyby**

Within the very limited development time (**14 months**) and budget (**a few M\$**), we could **demonstrated the capability** of this class of spacecraft to perform deep space mission by itself and it can be a useful tool of deep space exploration.

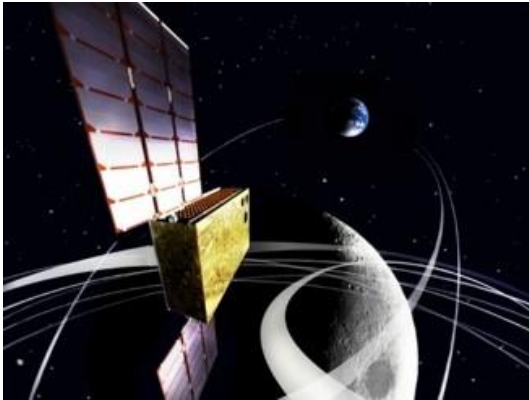


**PRYCYON proved technical readiness to deep space exploration by small satellite.**

# Next Challenges using SLS opportunity

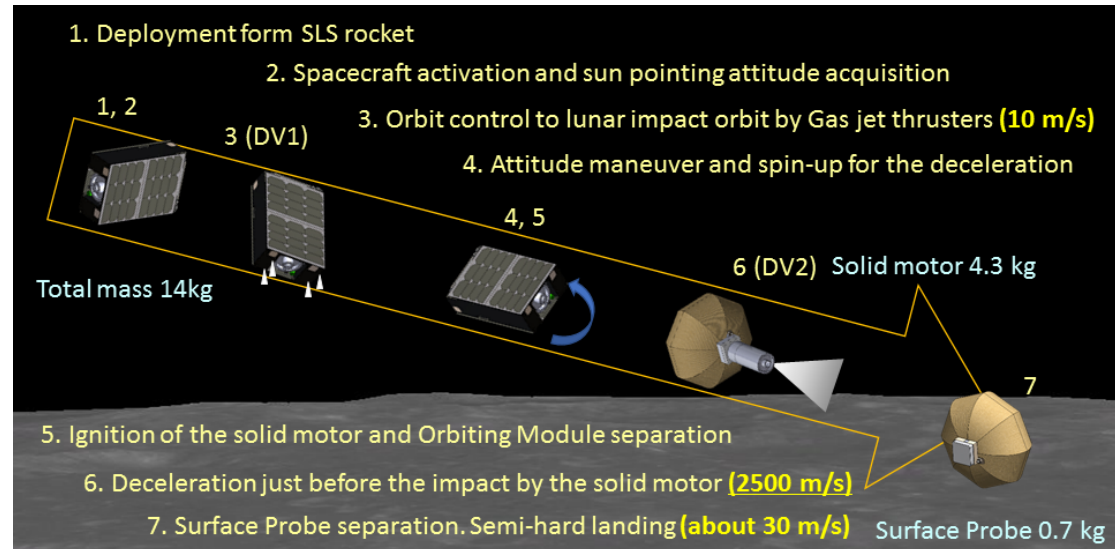
Two Japanese 6U satellites to challenge for planetary exploration are selected and developed as SLS secondary payload.

## EQUILLEUS & OMOTENASHI



**EQUILLEUS** (EQUilibrium Lunar-Earth point 6U Spacecraft) EQUILLEUS have been developed by JAXA and the University of Tokyo. EQUILLEUS aims to be the first CubeSat to Lunar Lagrange point using water resistojet thrusters.

**OMOTENASHI** has been developed by JAXA. OMOTENASHI aims to be the smallest lunar lander in world using retrojet(solid motor), airbag and crushable material.



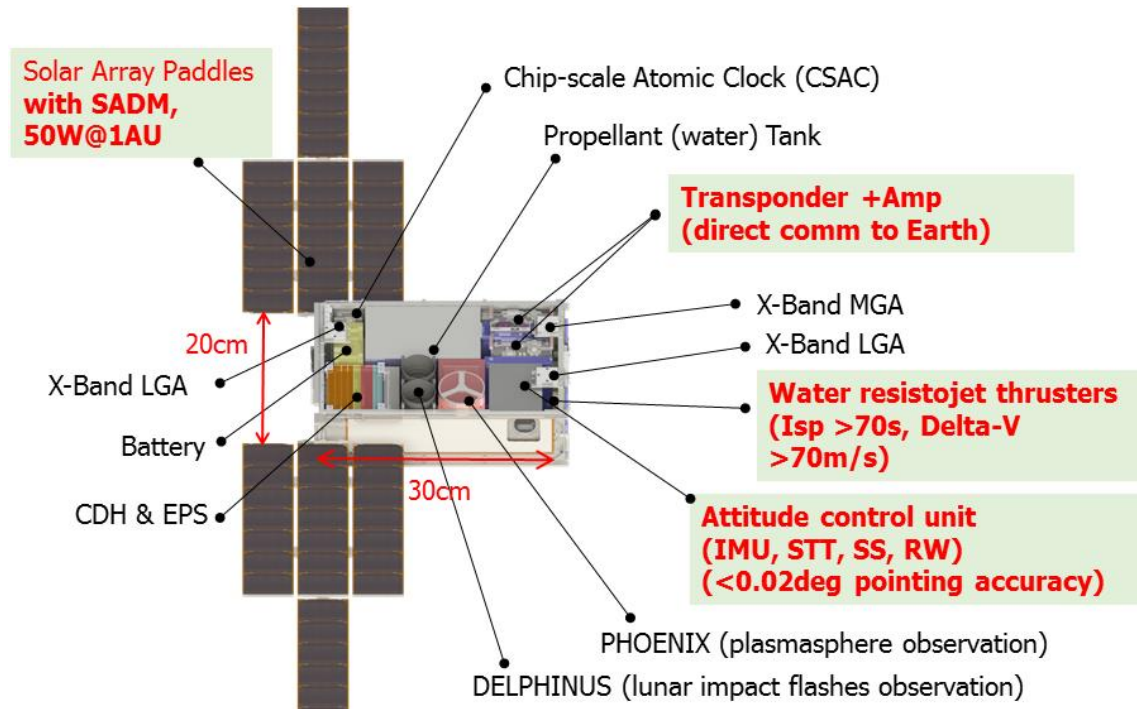
# Overview of EQUILLEUS

## [Engineering Objective] (primary mission)

Demonstration of **the trajectory control techniques within the Sun-Earth-Moon region** ( $< \sim 1.5M$  km from Earth) by a nano-spacecraft through the flight to the Earth-Moon Lagrange point L2 (EML2)

## [Science Objectives]

- \*Imaging observation of the Earth's plasmasphere
- \*Lunar impact flash observation
- \*Dust environment measurement in the cis-lunar region





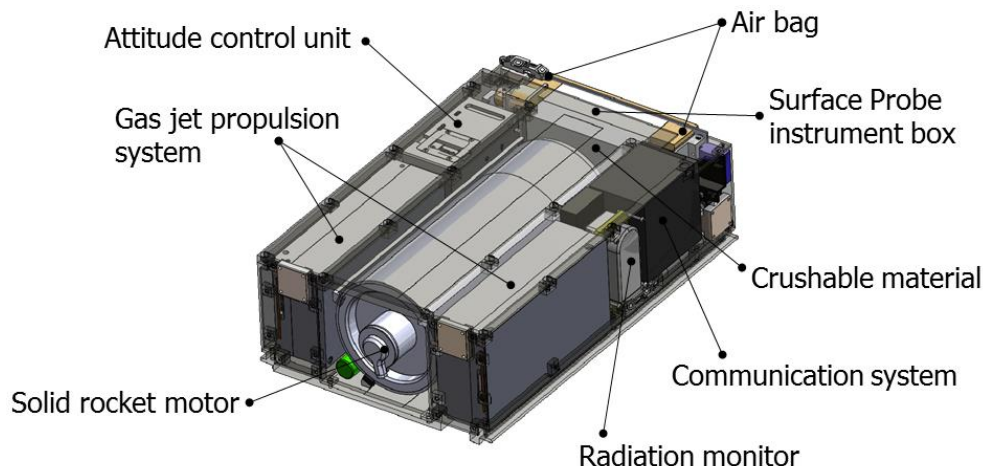
# Overview of OMOTENASHI

## <Objectives>

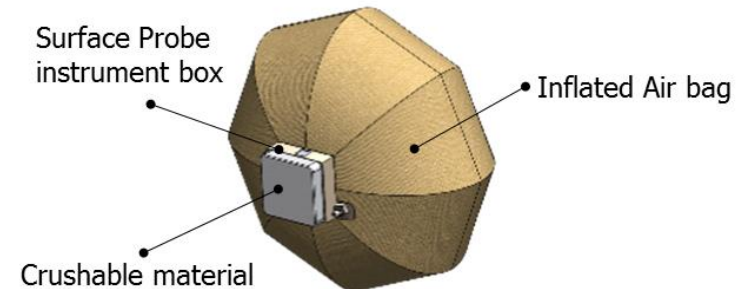
\*Development of **the smallest lunar lander** in the world and demonstrate the feasibility of the hardware for distributed cooperative nano-exploration system. Small landers will enable **multi-point exploration which is complementary with large-scale** human exploration system.

\***Observation of radiation** and soil environment of the moon surface by active radiation monitors and touchdown acceleration measurements.

### Launch configuration



### Landing configuration



# What is next step ?

In order to realize an attractive and valuable planetary exploration, “orbiters” and “landers” on a planets with gravity and atmosphere are necessary and indispensable.

## Can we realize “orbiters” and “landers” under the restricted resource constraint in small satellite ?

### \* **Lander (Entry, Descent, and Landing) technology**

Conventional EDL technique (ablator and parachute) required a large resource, high cost and complex systems.

→ Need innovative technique, like [a deployable aeroshell.](#)

### \* **Orbiter (orbit insertion) technology**

Conventional propulsion technique required a large resource.

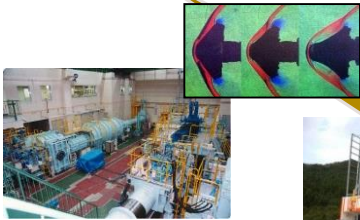
→ Need innovative technique, like [an aerocapture by a drag modulation using deployable aeroshell and jettison.](#)

**“Deployable aeroshell” is the final key technology to enhance a value of planetary exploration by small satellites.**

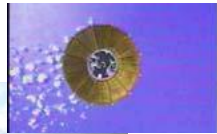
# Development of deployable aeroshell

Driven by The University of Tokyo and many Universities and JAXA/ISAS

2000 : fundamental study (wind tunnel test and numerical simulation)



2004 : 1<sup>st</sup> balloon drop test



2009 : 2<sup>nd</sup> balloon drop test



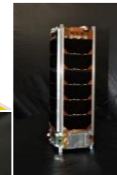
2012 : suborbital reentry test using sounding rocket.



2017 :  
Nano satellite EGG  
From ISS by JSSOD



2005- :  
Hypersonic wind tunnel test  
Full scale wind tunnel test



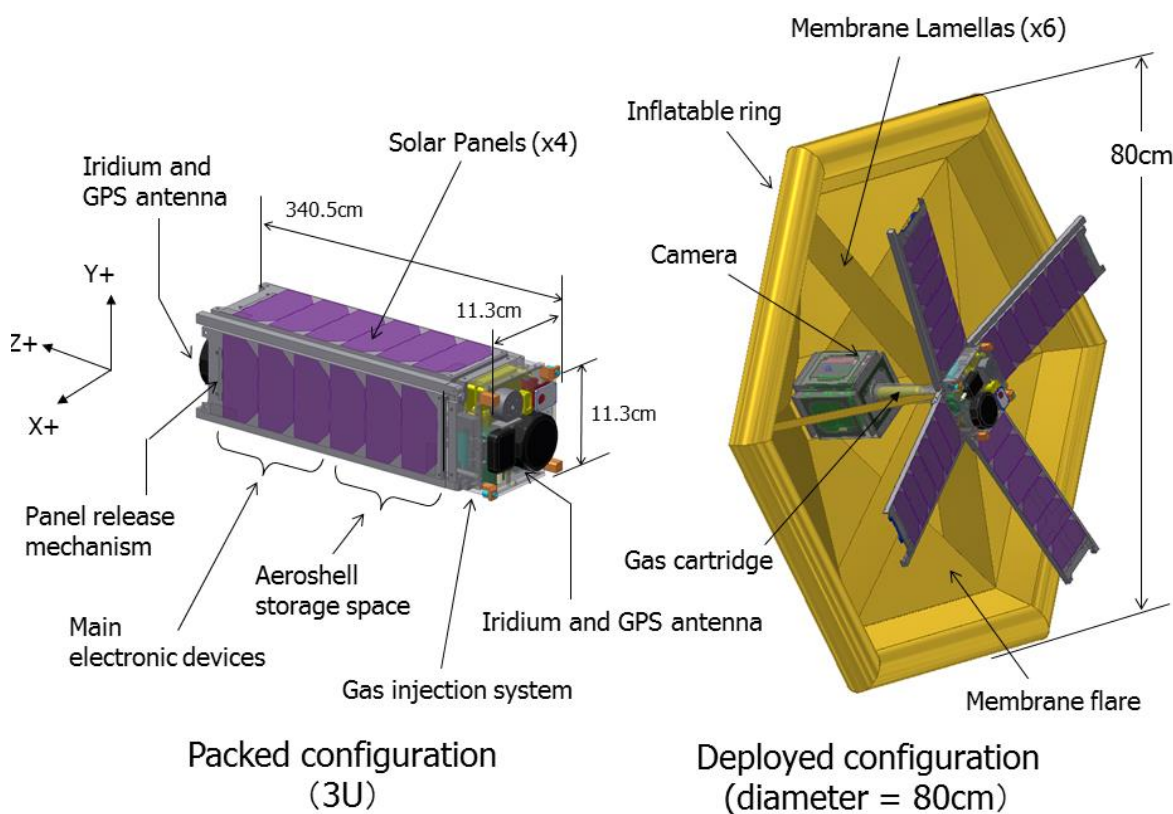
Apply to planetary  
Exploration  
by Small sat?

# Overview of EGG

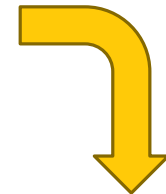
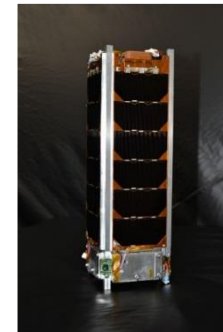
*"EGG" ~re-Entry satellite with Gossamer aeroshell and GPS/Iridium*  
 is a 3U-nanosatellite deployed from ISS using JSSOD system equipped in KIBO

## <EGG's objective>

To demonstrate indispensable technologies for a future atmospheric-entry flight test of the inflatable aeroshell in Low-Earth-Orbit

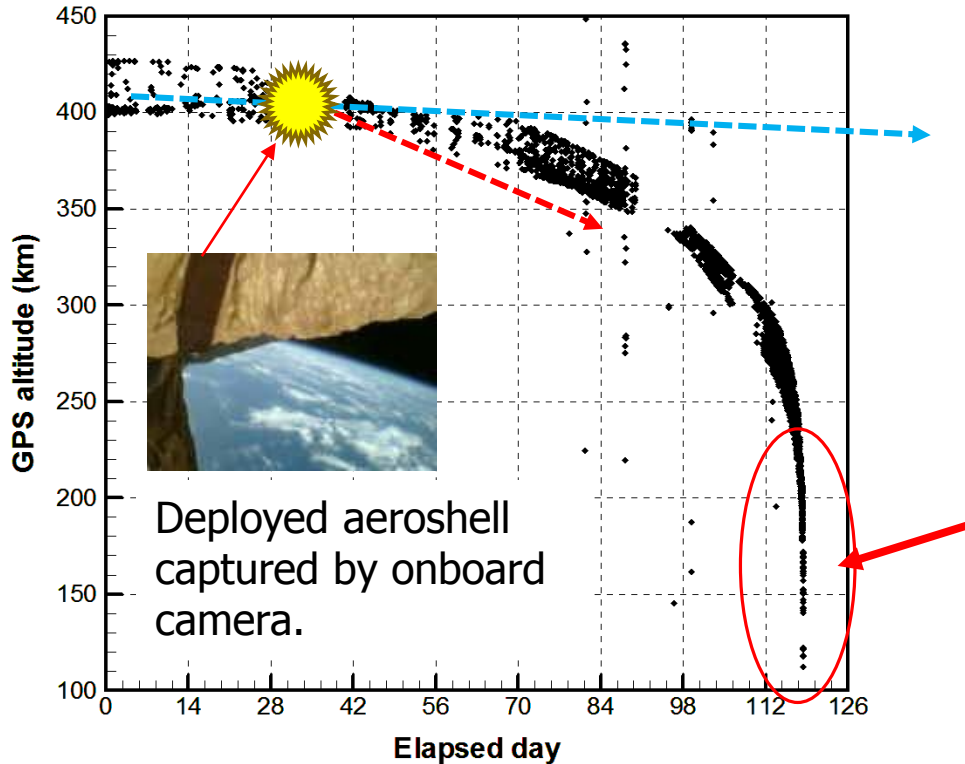


3U-size(10 x 10 x 30cm)  
 Mass : 4kg



# EGG' Achievements and to BEAK

Altitude history acquired by COTS GPS



- Demonstration of aeroshell deployment system installed in 3U nanosatellite in LEO.
- COTS GPS receiver (firefly) can identify EGG's position during flight and acquired the orbital decay process of EGG.
- Observation of orbital decay process due to aerodynamic force acting on the deployed aeroshell and reentry phase using Iridium SBD Tele-com system

## Next planned demonstrator utilizing EGG's Heritage

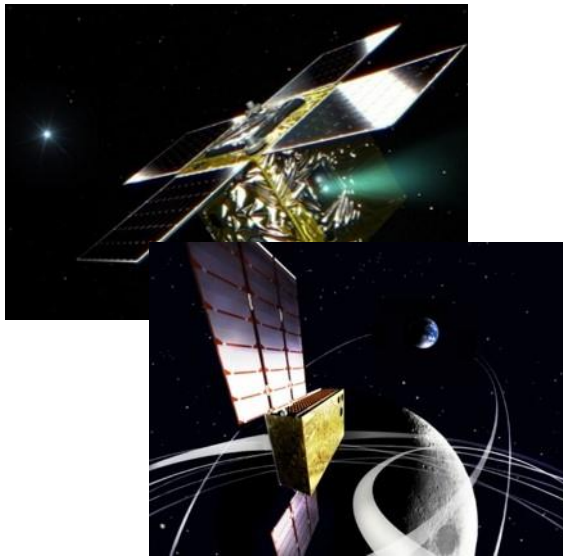
2020: BEAK (*Breakthrough by EGG-derived Aerocapture Kilt Vehicle*)

→ Aero-assisted orbit change of drag moderation will be demonstrated

**by a jettison of the deployable aeroshell in Earth's atmospheres.**

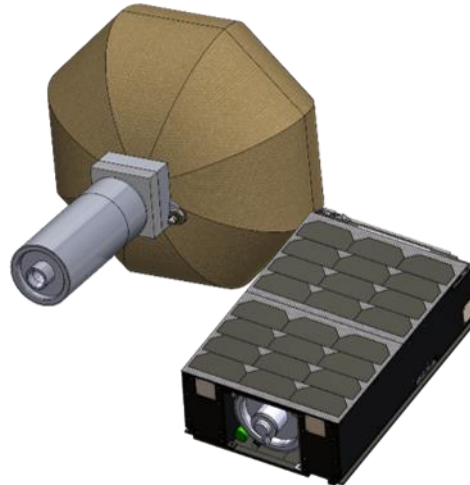
# Future vision

## PROCYON & EQUULEUS



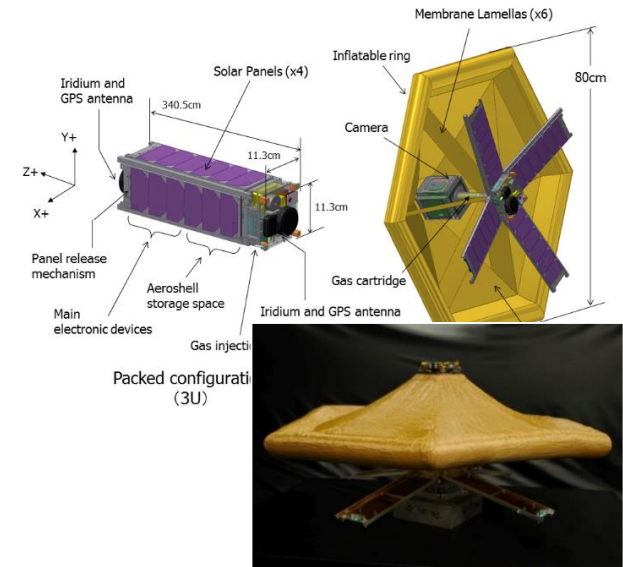
- \*Small sat spacecraft bus system technology
- \*Orbital maneuver technique with ion thruster and/or water resist jet

## OMOTENASHI



- \*Landing technology (air bag, retrojet, crushable)
- \*Solid motor technology for small satellite

## EGG & BEAK



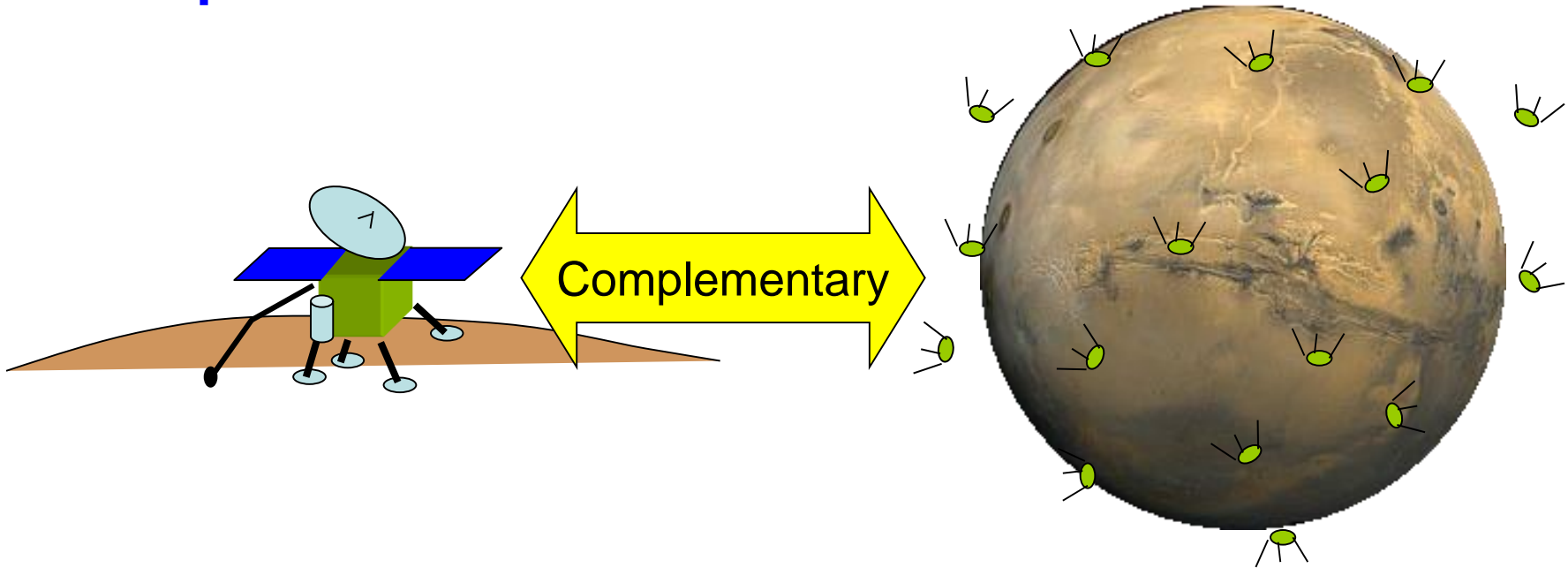
- \* Deployable aeroshell technique for atmospheric entry.
- \* Orbital insertion technique by drag modulation aerocapture

**Open a door of planetary exploration by small satellites**

# Paradigm shift in planetary exploration

## Single-Point-Type Exploration

## Network-Type Exploration By flock of scattered probes



\*Detailed and Precise information available

\*Mobility is a problem

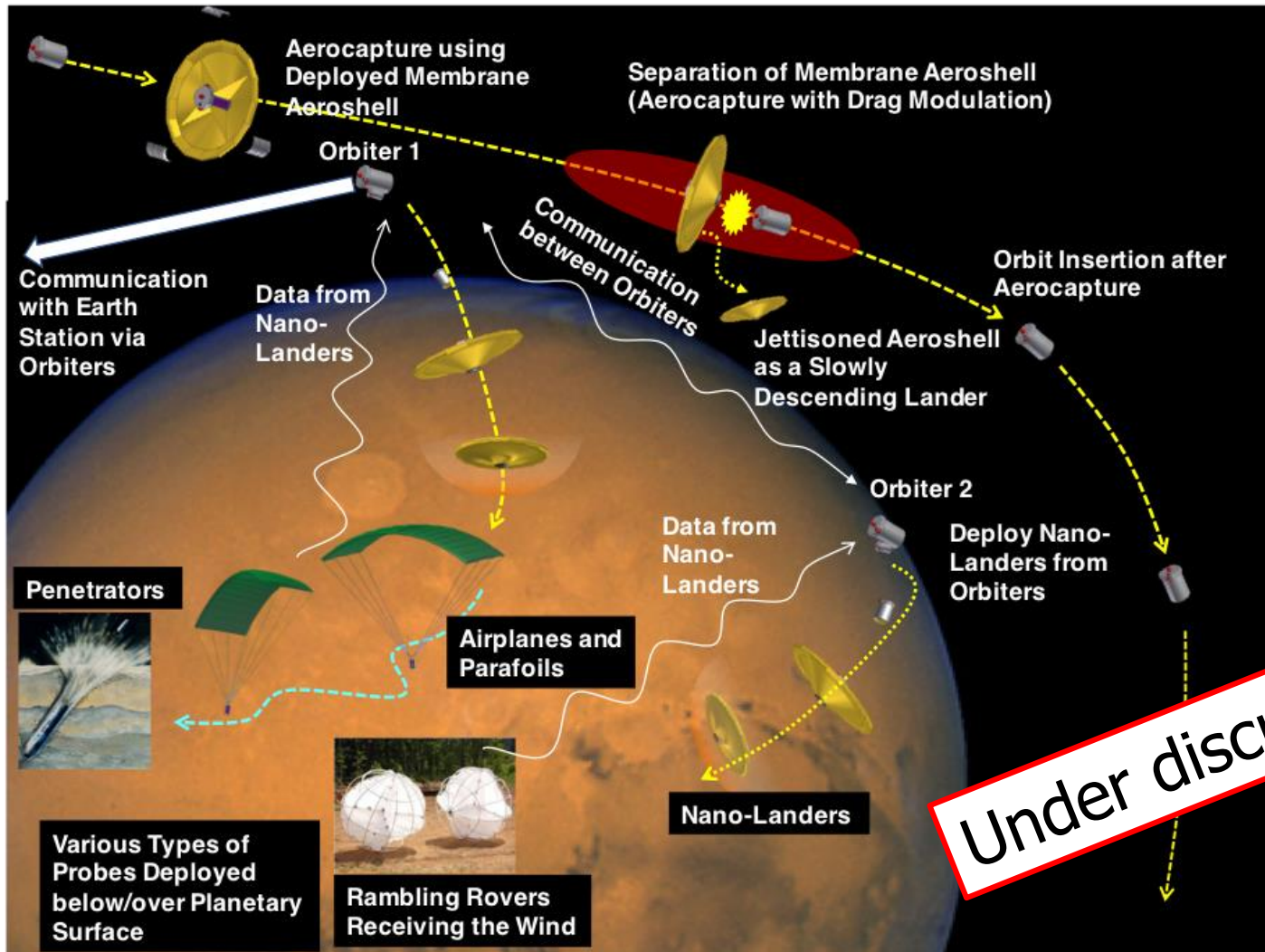
\*Hard to understand as  
"a **SYSTEM of complexity**"

Understanding of planet as "a **SYSTEM of Complexity**" based on synthesis of information obtained by simultaneous and Multipoint observation using a group of spatially dispersed probes.

**Small sat and nano-landers technique can realize this network exploration.**

# Proposed next projects

## ***SPUR*** *Scattered nano Probes Unfolded Reconnaissance*





# Summery

In Japan, some planetary exploration missions by small satellite-class is active on the heritage the cube-sat, hodoyoshi-sat and ISAS's small flight vehicle experiment (balloon and rocket) technology.

These projects are driven by collaborating with JAXA/ISAS, the University of Tokyo and many universities.

Small satellite technology has a potential as game changer for planetary exploration. For example, it may open a new world, that is, network exploration on the planet by scattered probes.